
Operational Aspects of Space Weather-Related Missions

Richard G. Marsden, ESA/SCI-SH

Outline

- ❑ SOHO: Example of Near-Earth Observatory-class Mission
- ❑ Ulysses: Example of Deep Space Monitor-class Mission
- ❑ Solar Orbiter: Example of Future Space Weather-related Mission
- ❑ Summary

SOHO

The Solar and Heliospheric Observatory (SOHO) is a cooperative mission between ESA and NASA to study the Sun, from its deep core to the outer corona, and solar wind.

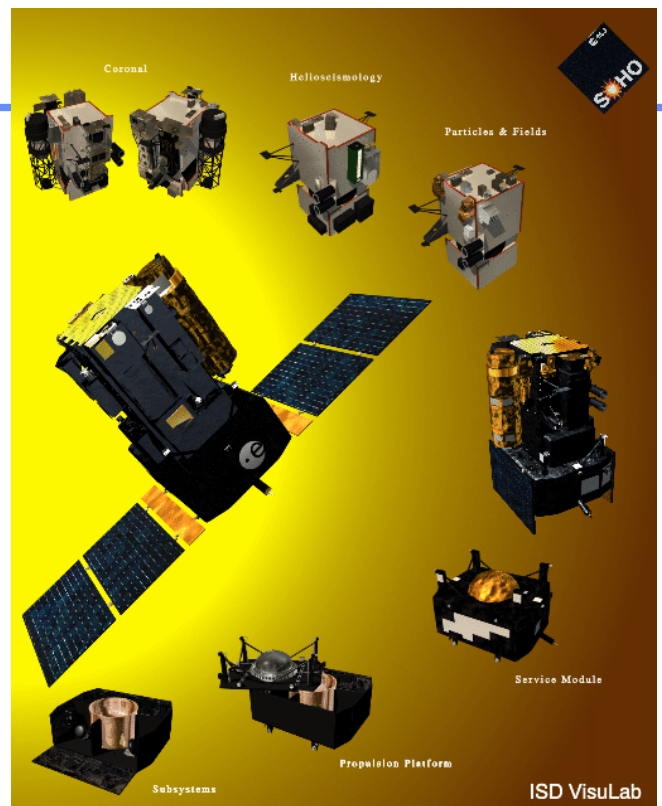
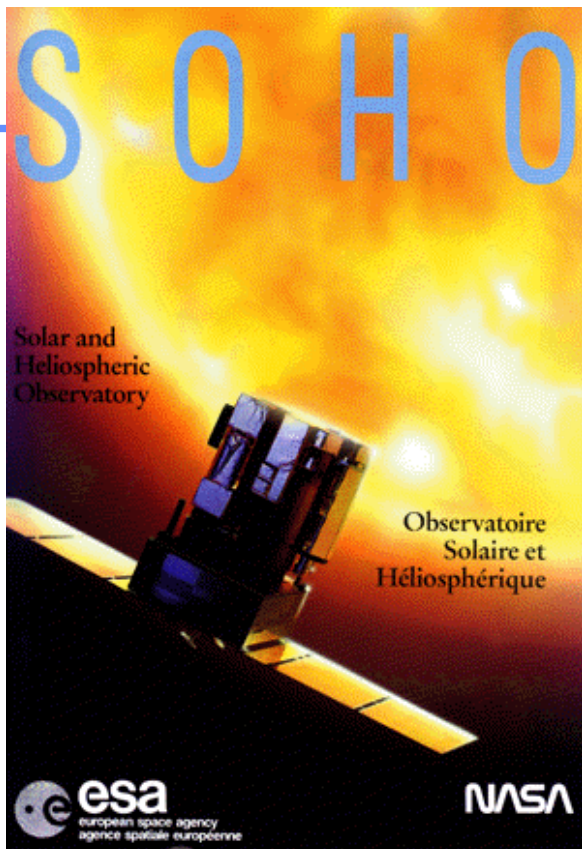
Key Mission Characteristics:

- ❑ Launch: Dec 1995
- ❑ Lifetime: approved until 2007
- ❑ Orbit: halo orbit around Sun-Earth L1 Lagrangian point
- ❑ Sun-pointing, 3-axis stabilized platform
- ❑ Payload: 12 sets of instruments (remote-sensing and *in-situ*)
- ❑ Tracking: DSN (3 x 1.6 hrs and 1 x 8 hr pass per day)
- ❑ Telemetry: 40 kbps (160 kbps for MDI high data rate mode during 8-hr pass)



Alpbach Summer School 2002

30 July 2002



Alpbach Summer School 2002

30 July 2002

SOHO Operations Concept

□ Near-Real-Time Operations:

➤ Observatory-class Mission

➤ Typical Time Delays:

○ Telemetry receipt by Ground Station => Experimenter (EOF) < 5 sec

○ Experiment Command from EOF => SOHO < 1 min

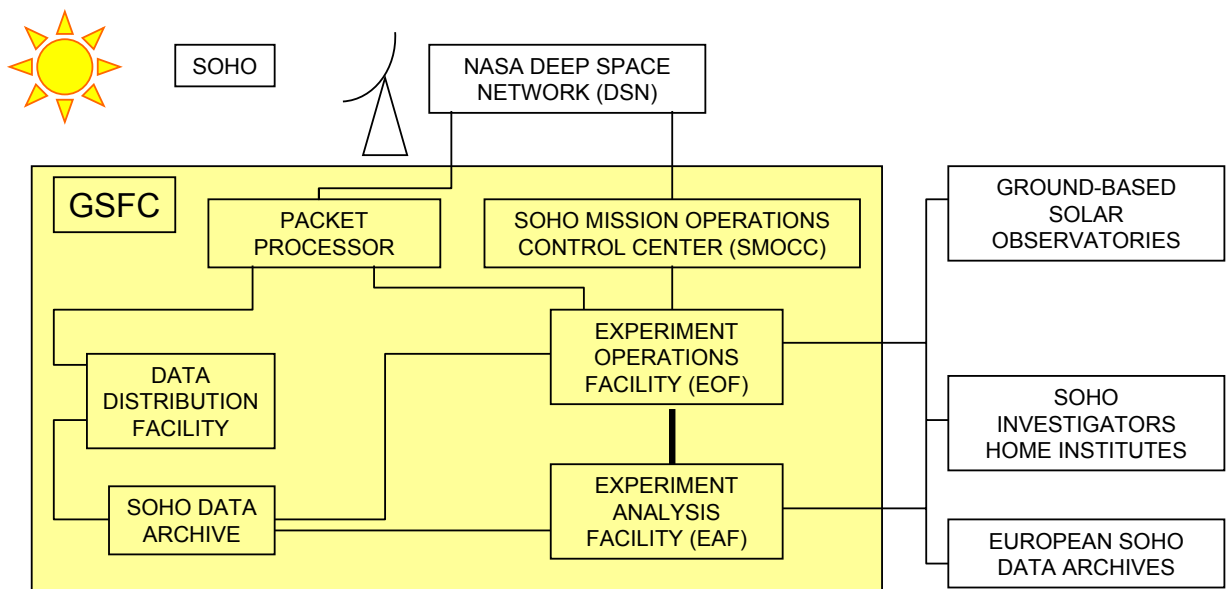
➤ EOF and EAF connected by 100 Mbps link

➤ PI Teams and Flight Ops Team Co-located

➤ Rapid Exchange of Complementary Data

○ Ground-based and other Space Observatories (e.g. TRACE)

SOHO Ground System



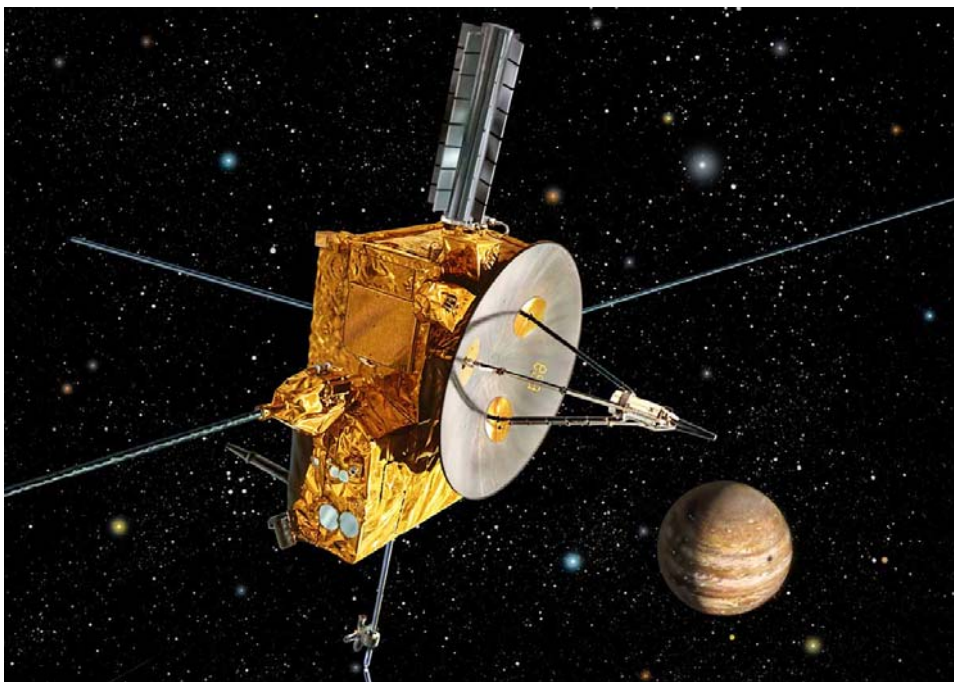
Ulysses

Ulysses is a cooperative mission between ESA and NASA to study the Sun and heliosphere in 3 dimensions.

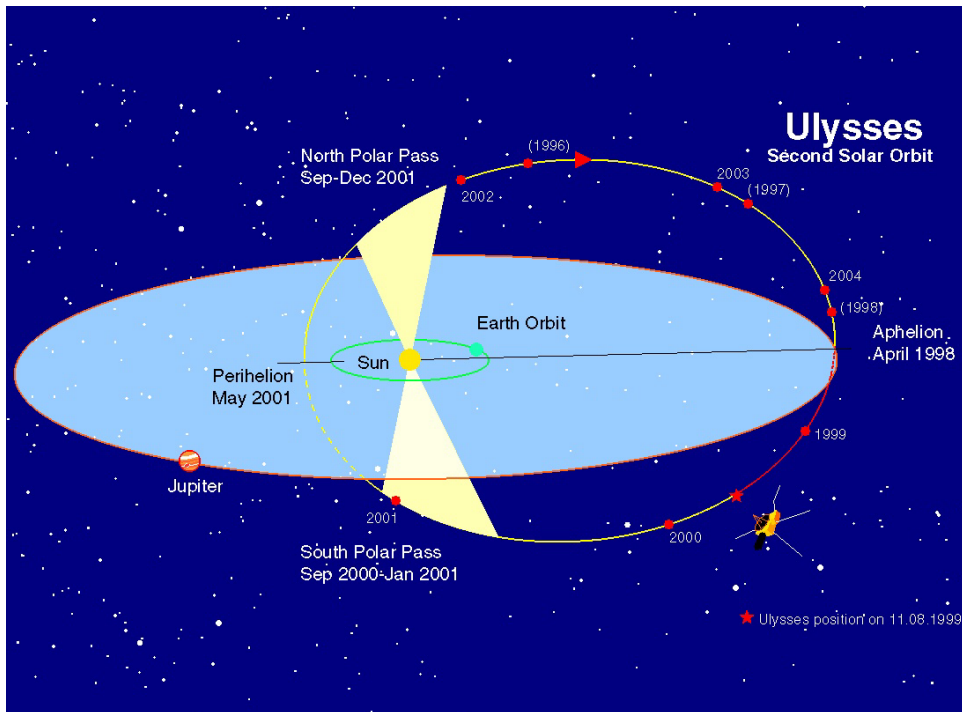
Key Mission Characteristics:

- ❑ Launch: Oct 1990
- ❑ Lifetime: approved until Sep 2004
- ❑ Orbit: heliocentric orbit inclined at 80° to the ecliptic plane
- ❑ Earth-pointing, spin-stabilized platform (5 rpm)
- ❑ Payload: 9 sets of instruments (*in-situ* particles and fields; gamma-ray bursts)
- ❑ Tracking: DSN (1 x 10 hr pass per day)
- ❑ Telemetry: 0.5 kbps (1 kbps real-time mode)

Ulysses



Ulysses

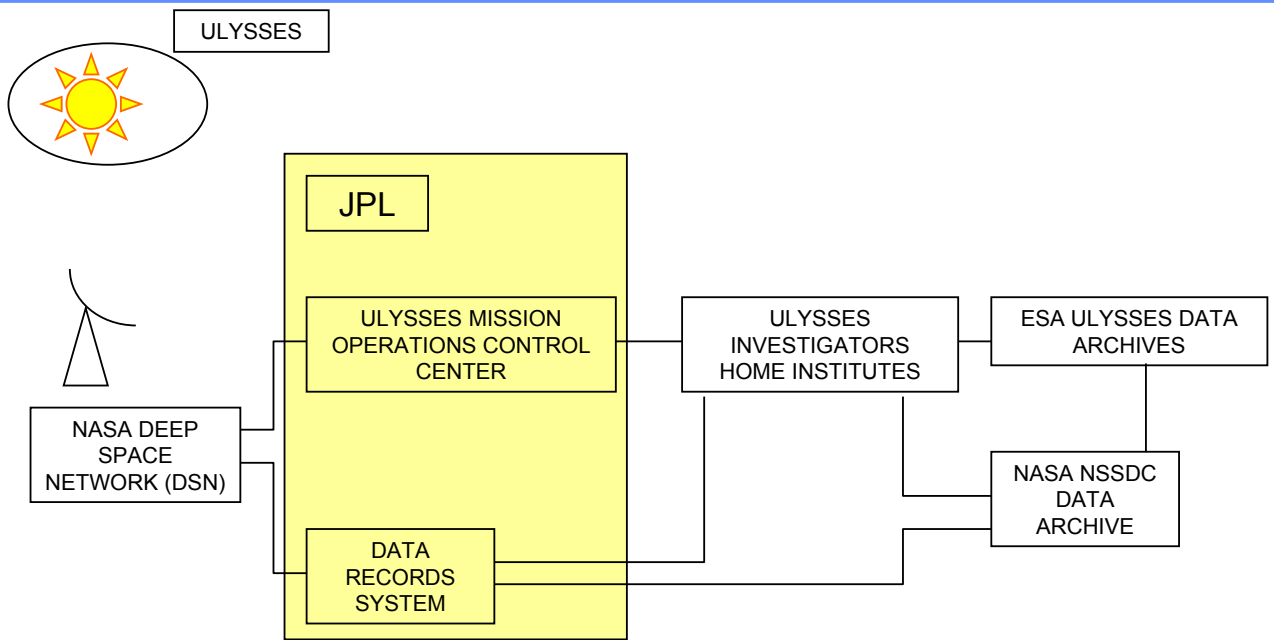


Ulysses Operations Concept

□ Non-Real-Time Operations:

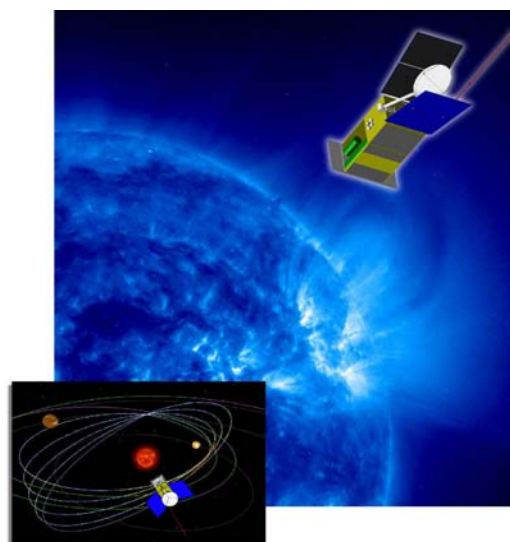
- Monitoring-class Mission
- No Ulysses EOF / EAF (Formal interface between PI Teams and Flight Ops Team is via e-mail and/or dedicated fax number)
- Spacecraft Operations Team (ESA personnel) located at JPL
- Typical Time Delays:
 - One-Way Light Time (OWLT) up to 52 min
 - Telemetry receipt by Ground Station => Experimenter
 - ◆ Quick-look data < 30 min
 - ◆ Final Experiment Data Record < 18 days
- Experiment Commanding via SOT (3-week lead time)

Ulysses Ground System



Solar Orbiter

A High-resolution Mission to the Sun and Inner Heliosphere



Solar Orbiter

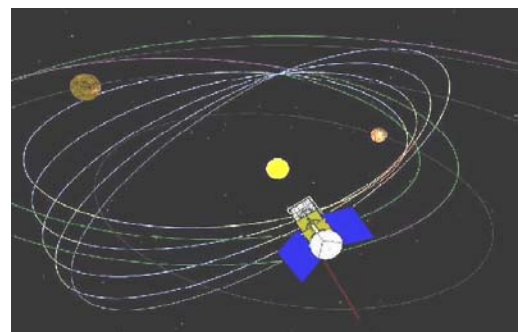
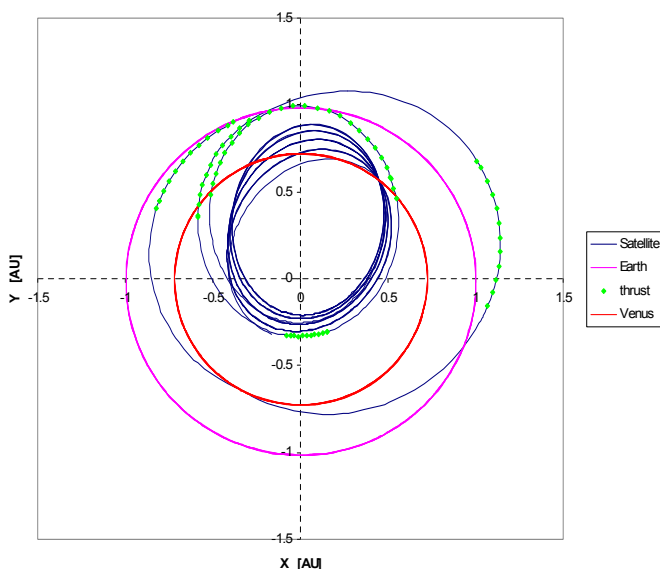
Solar Orbiter is a future ESA mission to study the near-Sun environment, using a combination of in-situ and remote-sensing observations. Solar Orbiter will also image the polar regions for the first time.

Key Mission Characteristics:

- ❑ Launch: 2011/2012
- ❑ Orbit: heliocentric with heliosynchronous phase (perihelion 0.21 AU) and high-inclination phase (up to 38° latitude)
- ❑ Sun-pointing, 3-axis stabilized platform
- ❑ Solar Electric Propulsion
- ❑ Payload: remote-sensing and *in-situ* instrument packages
- ❑ Operations: highly autonomous (esp. during perihelion passes)

Novel Orbital Design

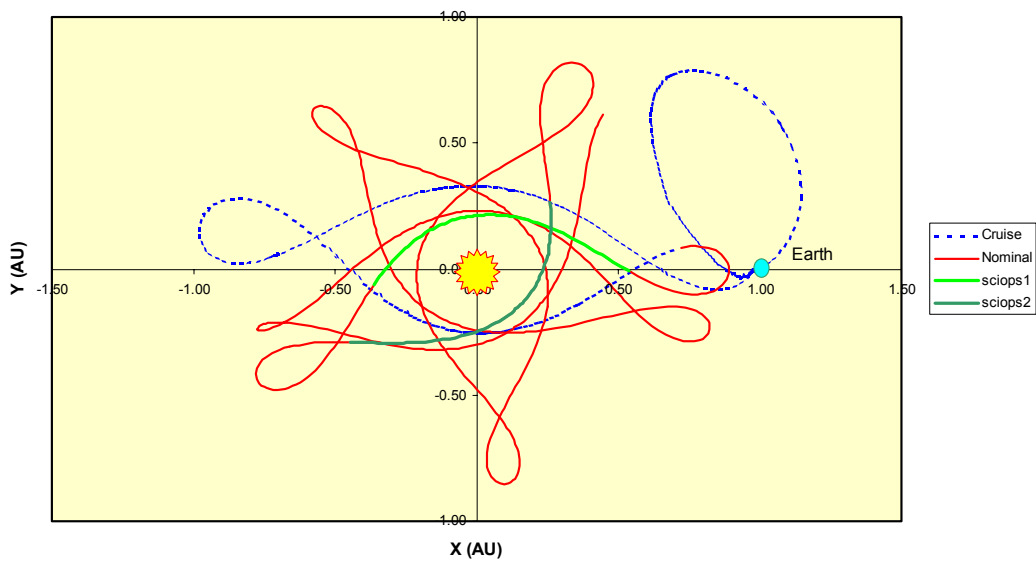
❑ Projected trajectory



- *closer to the Sun*
- *out of the ecliptic*

Fixed Sun-Earth View

Solar Orbiter trajectory in fixed Sun-Earth coords.



Nominal Mission

